



Benchmarking ALSSAT with ALiSSE : Aligning Life Support System Optimization with Ecosystem Efficiency

Chukwuemeka (Emeka) Ukaga, PE

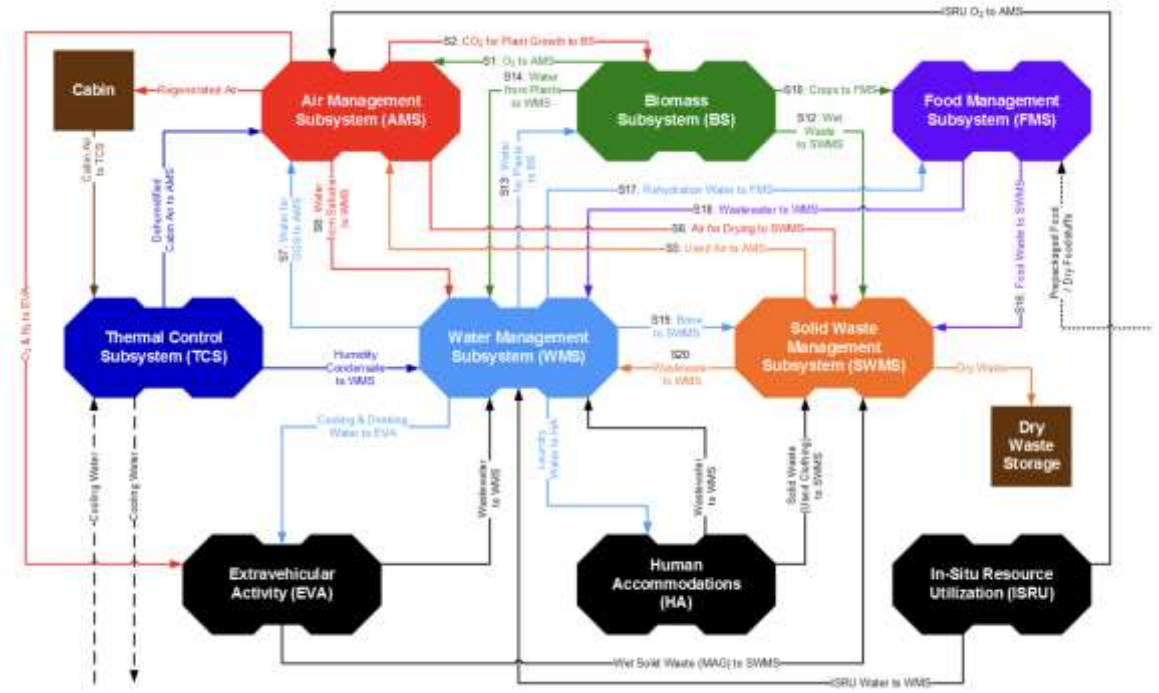
Executive MBA Space Architecture & Management Candidate

Academy for Continuing Education

TU Wien, Austria

- What is NASA's Advanced Life Support Sizing Analysis Tool?
- Use of Advanced Life Support System Evaluator Evaluation Criteria
- Hypothesis
- Methodology
- Findings / Results
- Analysis / Conclusions
- Open Questions & Next Steps

- First developed by the Crew and Thermal Systems Division (CTSD) of Johnson Space Center (JSC) in 1997
- Updated by Lockheed Martin in 2005 and then by Barrios Technology and Hamilton Sundstrand (now part of Collins Aerospace) again in 2012.
- Helpful for preliminary ECLSS design, developing trade studies, and analyzing system tradeoffs.



Overall ALSSAT Mass Balance — From User's Guide for the ALSSAT (NASA/TM-2017-219287)

1997

CTSD initial Development

2005

Lockheed Martin adds food, solid waste, and thermal control subsystems

2012

Barrios & HS upgrade entire ALS database, CO₂ calculations, and cabin air & food subsystem.

- Advanced Life Support System Evaluator (ALiSSE) defines seven key metrics to evaluate and compare life support system designs.
- Metrics capture the technical, operational, and ecological performance of closed-loop habitats, guiding trade-offs between resource use, safety, and autonomy.



Mass



Energy & Power



Efficiency



Crew Time



Risk for Human



Reliability



Sustainability

- Advanced Life Support System Evaluator (ALiSSE) defines seven key metrics to evaluate and compare life support system designs.
- Metrics capture the technical, operational, and ecological performance of closed-loop habitats, guiding trade-offs between resource use, safety, and autonomy.
- Three values **directly measured** by ALSSAT—two **partially measured**.



Mass



Energy & Power



Efficiency



Crew Time



Risk for Human



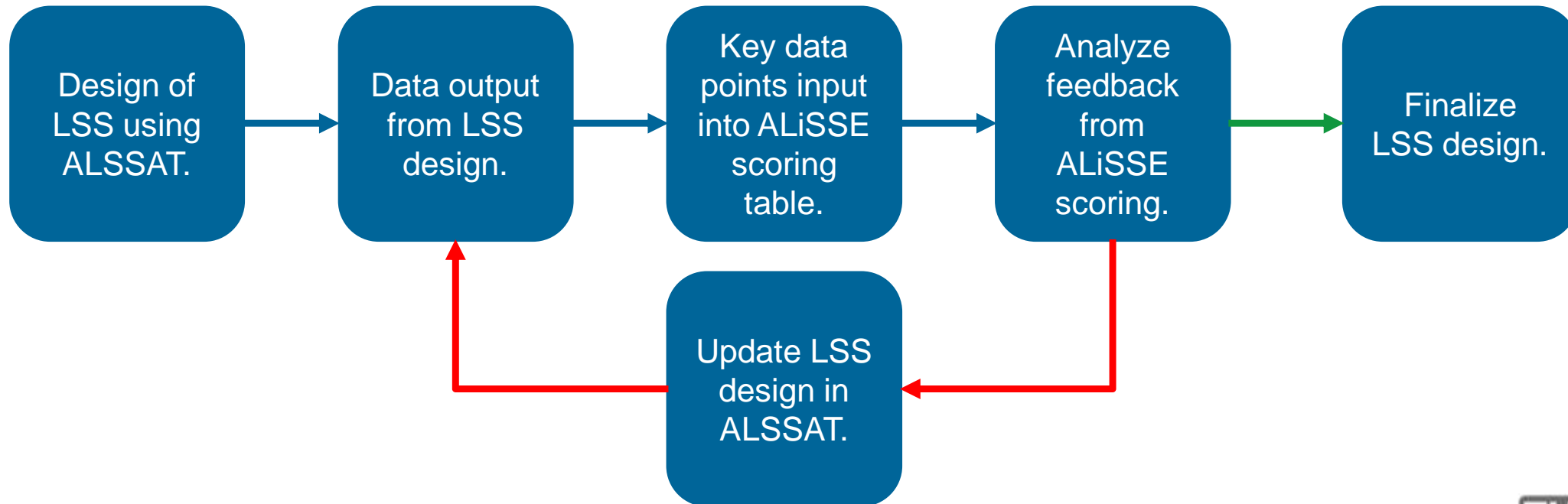
Reliability



Sustainability

Incorporating the ALiSSE evaluation criteria as benchmarking metrics during conceptual mission design will yield measurably different optimization outcomes — favoring higher ecological closure and system resilience even when ESM increases slightly.

- Designed a sample mission in ALSSAT.
- Nonplanetary; Crew of (4); Mission duration of (180) days
- No docking pressurization and (2) cabin repressurizations.
- Sized LSS based on criteria and evaluated with ALiSSE scoring table.



Findings – ALSSAT Snapshot

CDS		Case #1	Case 1	(Sample cas					
ALS SUBSYSTEMS		Tech./Data	Mass (kg)	Vol(m ³)	Power (W)	Cooling (W)	Crewtime, MMH/Dur	ESM (NCT), kg	ESM (CT), kg
Air			1,139.87	2.38	2,769.84	2,213.93	6.51	3,334.38	3,338.94
Biomass			0.00	0.00	0.00	0.00	0.00	0.00	0.00
Food			1,269.68	3.49	10.00	10.00	0.00	1,510.46	1,510.46
Waste			318.11	3.21	783.84	643.55	209.59	1,113.84	1,260.56
	SoIWC	Urinal & Commode	216.24	1.75	241.00	241.00	29.59	526.02	546.73
	SWPS		101.87	1.46	542.84	402.55	180.00	587.82	713.82
	SWTP	PWMC+ALS_Store	101.87	1.46	542.84	402.55	180.00	587.82	713.82
	SWPS_Tankage		0.00	0.00	0.00			0.00	0.00
	MCV							0.00	0.00
	Process_Control		0.00	0.00	0.00	0.00		0.00	0.00
	PQM		0.00	0.00	0.00			0.00	0.00
	SPD		0.00	0.00	0.00	0.00		0.00	0.00
Water			1,000.84	3.27	676.02	676.02	0.00	1,759.88	1,759.88
Thermal			397.75	0.90	3,156.69	3,156.69	0.99	2,982.54	2,983.23
Total 6 ALS Subsystems			4,126.24	13.25	7,396.38	6,700.19	217.09	10,701.11	10,853.07
EXTERNAL INTERFACES									
		Tech./Data	Mass (kg)	Vol (m ³)	Power (W)	Cooling (W)	Crewtime, MMH/Dur	ESM (NCT), kg	ESM (CT), kg
Extravehicular Activities			130.35	0.49	20.83	20.83	0.00	179.72	179.72
Human Accommodations			407.66	0.95	0.00	0.00	0.00	470.97	470.97
Total External Interfaces			538.01	1.44	20.83	20.83	0.00	650.69	650.69
ADDL. CONTINGENCY									
Total addl. contingency (not included above)			68.76	0.19	10.00	10.00	0.00	89.57	89.57
Total ALS Subsystems & Ext. Interfaces (w/Contingency)			4,733.01	14.88	7,427.22	6,731.03	217.09	11,441.36	11,593.32
Total ALS Subsystems & Ext. Interfaces (w/out Contingency)			4,664.25	14.69	7,417.22	6,721.03	217.09	11,351.80	11,503.76

Findings – ALiSSE Scorecard

- Applied a priority weighting towards **directly measured** values (e.g., Mass 30 %, Energy 25 %, Crew 25 %) and deprioritized value to **partially measured** values (e.g. Efficiency 10 %, Sustainability 10%)

Metric	Raw Value	Units	Lower Better?	Min (Baseline)	Max (Benchmark)	Weight	Normalized Value (0-1)	Weighted Score
Mass	4733.01 kg		Yes	3000	6000	0.3	0.422	0.127
Energy & Power	14.16 kW		Yes	8	18	0.25	0.384	0.096
Crew Time	217.09 hr		Yes	100	300	0.25	0.415	0.104
Efficiency	85 %		No	70	95	0.1	0.600	0.060
Sustainability	80 %		No	60	90	0.1	0.667	0.067
Composite Score	0.453003167						AVG: 0.498	

- $N = (X - \text{Min}) / (\text{Max} - \text{Min})$; (When a higher value is better)
- $N = 1 - (X - \text{Min}) / (\text{Max} - \text{Min})$; (When a lower value is better)
- Weighted score: $N_i \times w_i$
- Composite score: $\sum (N_i \times w_i) = 0.50$



Mass



Energy & Power



Efficiency



Crew Time



Risk for Humans



Reliability



Sustainability

Findings – ALSSAT Snapshot II

CDS		Case #1	Case 1	(Sample cas				
ALS SUBSYSTEMS								
	Tech./Data	Mass (kg)	Vol(m ³)	Power (W)	Cooling (W)	Crewtime, MMH/Dur	ESM (NCT), kg	ESM (CT), kg
Air		1,139.87	2.38	2,769.84	2,213.93	6.51	3,334.38	3,338.94
Biomass		0.00	0.00	0.00	0.00	0.00	0.00	0.00
Food		1,269.68	3.49	10.00	10.00	0.00	1,510.46	1,510.46
Waste		318.11	3.21	783.84	643.55	209.59	1,113.84	1,260.56
	SoIWC	216.24	1.75	241.00	241.00	29.59	526.02	546.73
	SWPS	101.87	1.46	542.84	402.55	180.00	587.82	713.82
	SWTP	101.87	1.46	542.84	402.55	180.00	587.82	713.82
	SWPS_Tankage	0.00	0.00	0.00			0.00	0.00
	MCV						0.00	0.00
	Process_Control	0.00	0.00	0.00	0.00		0.00	0.00
	PQM	0.00	0.00	0.00			0.00	0.00
	SPD	0.00	0.00	0.00	0.00		0.00	0.00
Water		1,000.84	3.27	676.02	676.02	0.00	1,759.88	1,759.88
Thermal		397.75	0.90	3,156.69	3,156.69	0.99	2,982.54	2,983.23
Total 6 ALS Subsystems		3,300.99	10.60	5,917.11	5,360.15	288.72	8,560.89	8,682.45
EXTERNAL INTERFACES								
	Tech./Data	Mass (kg)	Vol (m ³)	Power (W)	Cooling (W)	Crewtime, MMH/Dur	ESM (NCT), kg	ESM (CT), kg
Extravehicular Activities		130.35	0.49	20.83	20.83	0.00	179.72	179.72
Human Accommodations		407.66	0.95	0.00	0.00	0.00	470.97	470.97
Total External Interfaces		538.01	1.44	20.83	20.83	0.00	650.69	650.69
ADDL. CONTINGENCY								
Total addl. contingency (not included above)		68.76	0.19	10.00	10.00	0.00	89.57	89.57
Total ALS Subsystems & Ext. Interfaces (w/Contingency)		3,907.76	12.23	5,947.94	5,390.99	288.72	9,301.14	9,422.71
Total ALS Subsystems & Ext. Interfaces (w/out Contingency)		3,839.00	12.04	5,937.94	5,380.99	288.72	9,211.58	9,333.15

Findings – ALiSSE Scorecard II

- Reduction of crew from 4 to 3; mission days from 180 to 240.
- Energy and mass savings closer to 20%.
- Crew time increase tradeoff with less ESM and energy requirement.

Metric	Raw Value	Units	Lower Better?	Min (Baseline)	Max (Benchmark)	Weight	Normalized Value (0-1)	Weighted Score	
Mass	3300.99	kg	Yes	3000	6000	0.3	0.700	0.210	
Energy & Power	11.34	kW	Yes	8	18	0.25	0.670	0.168	
Crew Time	288.72	hr	Yes	100	300	0.25	0.060	0.015	
Efficiency	85	%	No	70	95	0.1	0.600	0.060	
Sustainability	80	%	No	60	90	0.1	0.667	0.067	
Composite Score	0.453003167							AVG: 0.539	

- $N = (X - \text{Min}) / (\text{Max} - \text{Min})$; (When a higher value is better)
- $N = 1 - (X - \text{Min}) / (\text{Max} - \text{Min})$; (When a lower value is better)
- Weighted score: $N_i \times w_i$
- Composite score: $\sum (N_i \times w_i) = 0.50$



- ALiSSE evaluation criteria can yield measurably different optimization outcomes. Very reliant on scoring mechanism used, mission priorities, and availability of confirmable data.
- While the updated mission would be less ideal from ESM benchmark, ALiSSE helped to highlight areas of optimization.
- Mass and Energy carried more weight for nonplanetary mission. Could vary for planetary missions (e.g. Risk and Efficiency).
- Missions where crew number less flexible may require other optimization techniques.
- ALSSAT used in early-stage mission design. Ideal time to evaluate alternative benchmarks for LSS design.

- Dynamic modeling & resource utilization.
- In-situ resource utilization.
- Mars Missions (longer travel distance, increased risk, what are changeable variables).

Thank You!

Questions?

emeka@ukaga.com

+1 (651)-335-9614

Vienna University of Technology / TU Wien

Academy for Continuing Education

Argentinierstraße 8, 1040 Wien, Austria